

## Re: Ballistic entry into circular orbit?

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- *From:* "Androcles" <[Headmaster@xxxxxxxxxxxxxxxx](mailto:Headmaster@xxxxxxxxxxxxxxxx)>
  - *Date:* Sun, 17 Feb 2008 07:04:41 GMT
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<nospam@xxxxxxxxxx> wrote in message  
[news:50hfr3pltum23ne824g02qdbq93thmbjf8@xxxxxxxxxx](mailto:news:50hfr3pltum23ne824g02qdbq93thmbjf8@xxxxxxxxxx)

| Hello all.

| Would someone please help me figure out the correct way to analyze a  
| problem? Namely, is it possible to literally shoot something into  
| circular orbit?

| Assuming a vacuum, basic orbital mechanics says the orbit intersects  
| the gun's location.

| Posit an object in a highly elliptical orbit about a point-like Earth  
| mass, perigee at the gun 4,000 miles out. To circularize the orbit at  
| 4,300 miles, we have to a) reduce the apogee from God knows what to  
| 4,300 miles and b) increase the perigee from 4,000 miles to 4,300  
| miles. Can we do that?

| One way to increase an orbit's perigee is to add energy, but you have  
| to do so when the satellite is at apogee. Obviously, you can't do  
| that by introducing atmospheric drag into the model because you'd be  
| subtracting energy, not adding it. Worse, the atmosphere is at the  
| wrong end of the orbit. On the other hand, I don't know if that's the  
| only way to increase an orbit's perigee.

| By the same token, you can remove energy when the satellite is at  
| perigee to reduce its apogee. That you can certainly do with  
| atmospheric drag. Unfortunately, while that'll pull the apogee down,  
| you're still stuck with a perigee at the gun. Again, I don't know if  
| that's the only way to futz with the apogee.

| For example, what happens if you add or subtract energy at a point in  
| the orbit somewhere between apogee and perigee? Can you thereby take  
| some apogee altitude and trade it for some perigee altitude?

| One thing I know for sure: any object 300 miles up will be in circular  
| orbit as long as it has 25,000 fps tangential velocity, zero radial  
| velocity, and is under no accelerations other than gravity.

## Re: Ballistic entry into circular orbit?

| The sixty-four million dollar question is: given a real launch angle and velocity, can you employ real atmospheric drag to make a ballistic projectile launched from the surface achieve that state?

| My first thought was to divide the muzzle velocity into radial and tangential components – or vertical and horizontal, near enough.

| Use drag and  $\sqrt{2gh}$  to figure the vertical component of the muzzle velocity needed to get the thing to 300 miles.

| Use drag and surface velocity at your latitude to figure the horizontal component needed to produce a 25,000 fps surplus after passing through the atmosphere.

| Muzzle velocity is then the root of the sum of the squares of the component velocities.

| Aim point altitude is the arctangent of the ratio of the component velocities, vertical to horizontal.

| Given strong enough materials, couldn't you thus put a projectile into just about any dynamic state you wanted to?

| BUT

| Orbital mechanics says the perigee will intersect the gun.

| I can't see how to reconcile those two ideas. One of 'em is obviously wrong, or at least incomplete.

| Any help with this would be appreciated; thank you.

| --

| Dave Typinski

It cannot hit the gun if the gun recoils, which it will by Newton's third law.

Perhaps this is a better way, it gets over the atmosphere problem:

<http://video.google.co.uk/videoplay?docid=2995709441588134017>

However, there is another consideration (or more).

1) The planet or moon will turn beneath the orbit, so that will move the gun away but 3 orbits later a mountain replaces the gun. Better start from the top of the highest peak.

2) Another body:

<http://faculty.ifmo.ru/butikov/Projects/Collection.html>

There is no solution to the three body problem.

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