

# Re: Lowest orbit and DeltaV

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On Jun 12, 8:20 am, "Jim Relsh" <jre...@xxxxxxxxxxxxxxxxxx> wrote:

"Jeff Findley" <jeff.find...@xxxxxxxxxxxxxxxxxx> wrote in message

[news:6d45b\\$484feaac\\$927a2cda\\$8517@xxxxxxxxxxxx](mailto:news:6d45b$484feaac$927a2cda$8517@xxxxxxxxxxxx)

"Jim Relsh" <jre...@xxxxxxxxxxxxxxxxxx> wrote in message  
[news:47f76\\$484fa12a\\$23972@xxxxxxxxxxxxxxxxxxxxxxxx](mailto:news:47f76$484fa12a$23972@xxxxxxxxxxxxxxxxxxxxxxxx)

What's the lowest practical orbit for a satellite to be in? I think I read somewhere that it was about 150km or so, but I keep wondering how they determine this. And what is the corresponding DeltaV to get into that orbit?

Not to sound like President Clinton, but that all depends on how you define the word practical.

The ballistic coefficient of an object in a very low earth orbit has a huge impact on how long it takes that object's orbit to decay. You could determine the minimum orbital altitude which would keep a one ton sphere of depleted uranium in orbit for a day, but is that really practical? Who would orbit such a thing for such a short period of time?

OK, let's assume I'm trying to put a mass of 1kg with the shape and size of a carton of milk in orbit for a minimum duration of 2 years. What altitude and DeltaV would I need for this:? Is it enough to know altitude and

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calculate DeltaV with the webform at:<http://home.att.net/~ntdoug/UCM2.html>.  
How would I determine the altitude?

\*\* Posted from <http://www.teranews.com> \*\*

That is a orbital velocity calculator and not a delta V calculator for launch into orbit.

You won't get a quick and easy answer, it would take some number crunching

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